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# Slave Finite Element for Non-Linear Analysis of Engine Structures

Volume II—Programmer's Manual and User's Manual

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## SLAVE FINITE ELEMENT FOR

## NON-LINEAR ANALYSIS OF ENGINE STRUCTURES PROGRAMMER'S MANUAL

## 1.0 INTRODUCTION

The programming aspects of SFENES are described in the User's Manual. The information presented here is provided for the installation programmer. It is sufficient to fully describe the general program logic and required peripheral storage. All element generated data is stored externally to reduce required memory allocation. A separate section is devoted to the description of these files thereby permitting the optimization of I/O time through efficient buffer description. Individual subroutine descriptions are presented along with the complete Fortran source listings. A short description of the major control, computation, and I/O phases is included to aid in obtaining an overall familiarity with the program's components. Finally, a discussion of the suggested overlay structure which allows the program to execute with a reasonable amount of memory allocation is presented.

#### 2.0 GENERAL PROGRAM LOGIC

The general organization of the SFENES program is illustrated in Figures 1 thru 6. The element routines consists of a control program and four principal computation phases; namely,

- 1. Input interpretation phase
- 2. Spatial integrations and computations
- 3. Matrix computations and time assembly
- 4. Stress recovery and element output phase

## 2.1 Control Program (Figure 1)

Routine "INTRFC" controls the element computation flow. Initially, the "Input Phase" and "Spatial Computational Phase" are executed. These two phases are outside of the iterative phases and need only be executed once. Routine "INTRFC" then executes the "Matrix Computation and Temporal Assembly Phases". Control is referred to the "MAIN" program for all spatial assembly and displacement computations,  $[K_F]\{u\}=\{F_D\}$ . The "INTRFC" portion is reentered for convergence checking and the stress recovery phase is then executed. The iterative process is repeated until convergence or the maximum number of iterations is achieved. The Control program is the interface between the element routines and the MAIN program unit.

## 2.2 Input Interpretation Phase (Figure 2)

The purpose of this phase is to read the formatted element input data and control data from the input file "IOIN". The data, in some cases (element loads and local grid coordinates), is used in additional computations and is finally output to binary files "IOEL", "IOLD", and "IOUS" for use by the remaining phases. This phase is only entered prior to the first iteration of the problem. Routine "INPUT" reads the formatted data, provides an echo print for the user and outputs all data to the output files. "GRIDG" computes all of the remaining grid point coordinates from local grid points 13-16 and the

corner thicknesses. Routine "LOADIN" reads the input element load data and computes the resulting grid point forces at the user input time increments.

## 2.3 Spatial Integrations and Computations (Figure 3 & 3A)

The "Spatial Computation Phase" is executed once prior to the first iteration. The function of this phase is to perform all spatial computations and any other "one" time computations which required spatial integrations. The primary tasks of this phase are outlined briefly below.

- a) The volume, area, and surface definite integrals are computed for the polynomial terms  $x^Ny^mz^1$  and  $\rho^kx^iy^jz^1$ ;  $0 \le n \le 9$ ,  $0 \le n \le 9$ ,  $0 \le 1 \le 7$ ,  $0 \le i \le 7$ ,  $0 \le j \le 7$ . These integrals are computed by routines VOLCOF, COLXYZ, CEAXYZ, CNLXYZ, CURCOF, CURPSI, CNRPSI, CURXYZ, CNRXYZ, TRLCOF, VOL-, VL-, REA-, RA-, SUR-, and SUP-.
- b) Routine LOADD computes the element force vector  $\{F\}$  by combining the input element loads as a function of time, (file IOLD) the appropriate spatial integrals from (a) above, the Hermitian time functions, and integrations over the local time ranges. The resulting combination of element body force, pressure and traction forces are output to file "IOFO" for intersolution of  $[K_F]\{u\}=\{F\}$  equation.
- The final function of this phase is to compute and store the spatial matrix partitions of [M], [K], [X], [A], [B], [C<sub>e</sub>], [B<sub>28</sub>], [P<sub>28</sub>], [C], [N<sub>x</sub>], and  $[P_{28}]^*[B_{28}]^{-1}$ . These matrices are stored on file "IOCF", "ICPR", and "IOTP" for use by the third phase. The matrices computed in this task are reused in subsequent iterations since the shape functions are constant.

## 2.4 Matrix Computation and Time Assembly Phase (Figure 4)

The function of this phase is to compute the element stiffness and stress matrices for each local time range. This involves the computation of the  $[E_6]$ 

matrices as a function of time, (routine EPQP19) and integrating over the local time intervals. The result of the time integrations is multiplied by the spatial matrices as computed in the spatial computation phase and assembled into the element  $[K_e]$ ,  $[S_e]$ ,  $[A_e]$  and  $[B_e]$  matrices for each local time interval. These computations may be made for either the linear elastic quad plate or the non-linear elastic-plastic quad plate elements (routines LEQP and EPQPB respectively). Spatial matrices are input from files "IOPR" and "IOTP" and prior iterative values of  $\{w\}$  and  $\{\sigma\}$  are input from file "IOUS". The "IOAC" and "IOAl" are used alternately to read the previous and store the current coefficients of polynomials in (t) used to compute the  $[E_7]$  matrices. File "IOKE", "IOAE", "IOBE", and "IOLD" are output files for storing element  $[K_e]$   $[A_e]$ ,  $[B_e]$  and  $[S_e]$  matrices respectively.

Routines "TBASEM" and "TABSEM" perform the temporal assembly of the element  $[K_e]$  matrices for the "LEQP" and "EPQP" elements respectively. Files "IOKE", "IOCE", "IOBE", "IOAE" control the input to these routines and the final element  $[K_f]$  matrices are output to file "IOKF". For the "EPQP" element the computation  $[B_e] \cdot [C_e]^{-1}$  is stored on file "IOLD" for element stress computations.

## 2.5 Stress Recovery and Element Output (Figure 5)

The "CNVRG" routine checks for convergence of all displacement/velocity values and stores the latest value of {u} on file "IOUL" for the next iteration. Element stress is recovered in routines "CSIGO" or "CSIGL" for element types "EPQP" and "LEQP" respectively. The input to these routines is contained in files "IOTF", the current displacement/velocity values {u}, and "IOLD", the stress matrices [S] or  $[C_e]^{-1}$  [B<sub>e</sub>]. The displacement/velocity {u} and stress { $\sigma$ } vectors are output to file "IOUS" to be used in the computation of [E<sub>7</sub>] for the next estimate of [K<sub>e</sub>]. Printed output of selected grids displacement and/or stress values for each iteration is provided for by

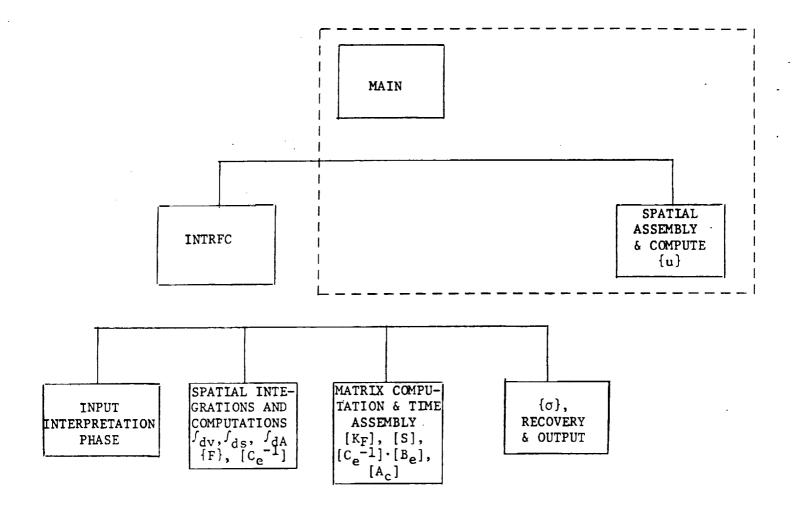
routine "WRITO". This output is at the major time intervals only. Convergence of the problem initiates the extrapolation and output of displacement/velocity values at user specified time intervals within the major time increments.

## 2.6 Displacement Computation (Figure 6)

Control is returned to the "MAIN" program after the stiffness  $[K_F]$  and stress [S] or  $[C_e^{-1}]^{\bullet}[B_e]$  matrices have been time assembled in the matrix computation phase. The "MAIN" program calls routine "COMPU" which performs the following functions:

- a) Read the partitions of the  $[K_f]$  matrix (see Figure 7 to Figure 9 for description of  $[K_f]$  matrix)
- b) Applies the user input boundary conditions to the individual partition  $\hbox{(eliminates rows and columns) and forms the reduced } [K_F] \ \hbox{matrix.}$
- c) Eliminates all rows containing displacement related data at times T=0 and  $T=t_N$  (rows 1-84 and rows 1N\*168+1 to N\*168+84 of the [K<sub>F</sub>] matrix).
- d) Reads the  $\{F_F\}$  vector partitions and time assembles the local partitions (see Figure 14 for a description of the assembled  $\{F\}$  vector).
- e) Applies user input boundary conditions as in step b) and c) above to the  $\{F_{\mathbf{F}}\}$  vector rows.
- f) Computes  $\{u\} = [K_F^{-1}]$   $\{F\}$  and stores data or file "IOTP" for stress recovery and output phase.

The newly computed displacement vector {u} is then read by the stress recovery phase and the next iteration is entered if required.



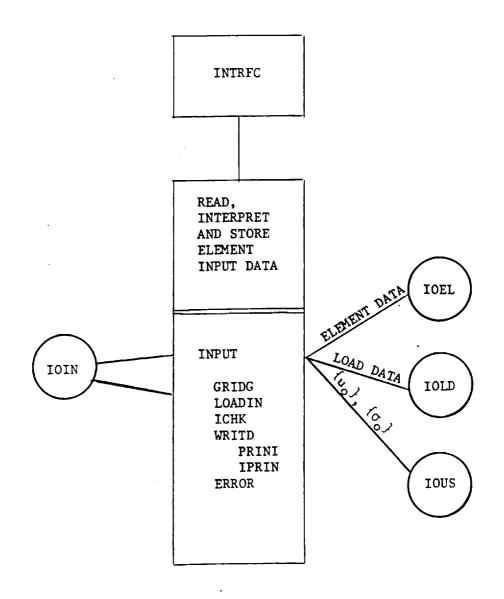


FIGURE 2

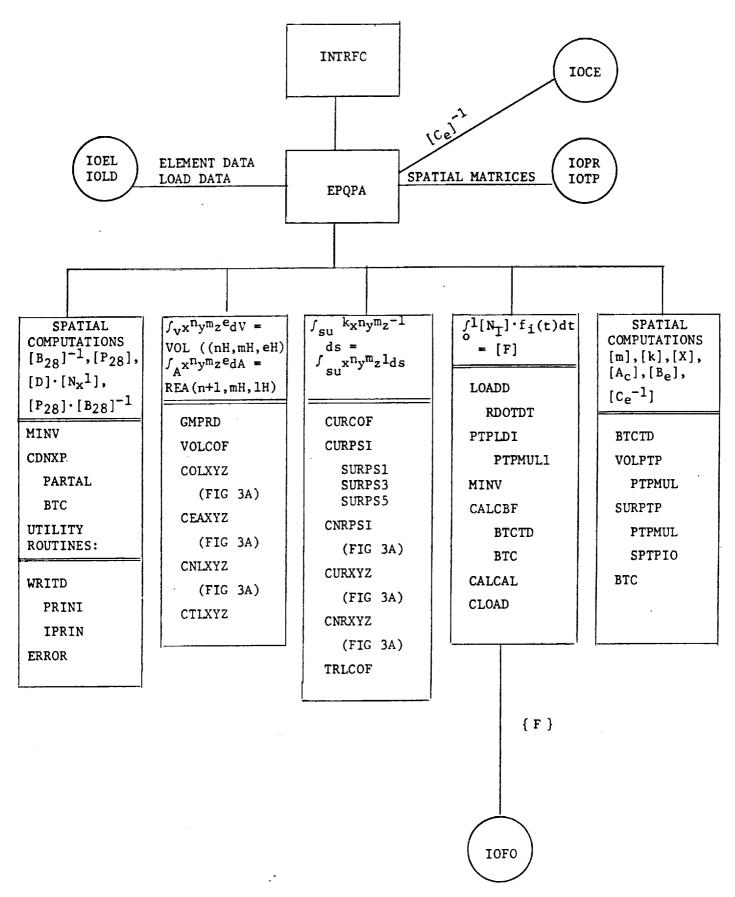


FIGURE 3

COLXYZ	CEAXYZ	CNLXYZ	CNRPSI	CURXYZ	CNRXYZ
$\int_{\mathbf{v}} \mathbf{x}^{\mathbf{n}} \mathbf{y}^{\mathbf{m}} \mathbf{z}^{1} d\mathbf{V}$	$\int_{A} x^{n} y^{m} z^{1} dA$	$\int_{\mathbf{v}} \mathbf{x}^{\mathbf{n}} \mathbf{y}^{\mathbf{m}} \mathbf{z}^{1} d\mathbf{v}$	$\int_{ds} p^k z^1 dS$	$\int_{ds} x^n y^m z^1 dS$	$\int_{ds} n^n y^m z^1 dS$
v	A	V	as	ds	ds
VOL111	REAll1	VOL181	SURPIX	SURXY1	SURX81
VOL211	REA211	thru	SURPY1	SURXY3	SURX91
VOL311 -	REA311	VOL011	SUP53X	SURX13	SURX01
VOL411	REA411	(13 Rout)	SUP63X	SURX23	SUR8Y1
VOL511	REA511		SUP73X	SURX33	SUR9Y1
VOL611	REA611	VOL183	SUPY13	SURX 43	SUROY1
VOL711	REA711	thru	SUPY23	SURX 53	SURX83
VOL113	REA113	VL1043	SUPY33	SURX 63	SURX93
VOL213	REA213	(32 Rout)	SUPY43	SURX73	SURX03
VOL313	REA313		SUP55X	SURXY5	SUR8Y3
VL313	REA413	VOL145	SUP65X	SURX15	SUR9Y3
VL363	RA413	thru	SUP75X	SURX25	SUROY3
VOL413	RA433	VOL845	SUP155	SURX35	SURX85
VL413	REA513	(29 Rout)	SUP165	SURX45	SUR8Y5
VL453	RA513		SUP175	SURX 55	SURX17
VOL513	RA553	VOL117	SUP255	SURX 65	SURX 27
VL513	REA613	thru	SUP265	SURX75	SURX37
VL543	REA713	VOL447	SUP275		SURX 47
VL553		(4 Rout)	SUP355		
VOL613	•		SUP365		
VOL713			SUP375		,
VL713	••		SUP455		
VL733			SUP465		
VOL115			SUP475		
VOL215			SURPS7		
VOL315					
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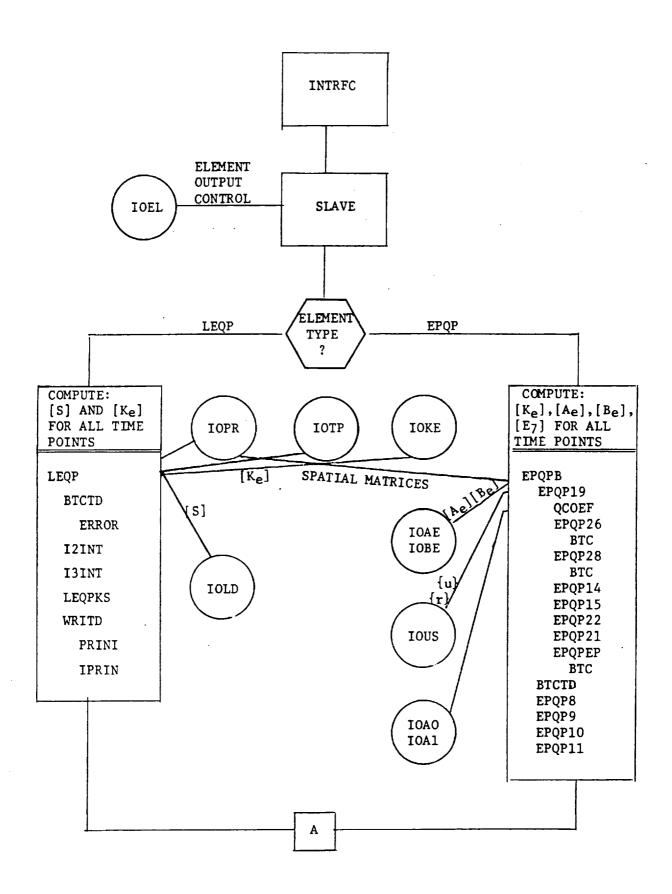


FIGURE 4

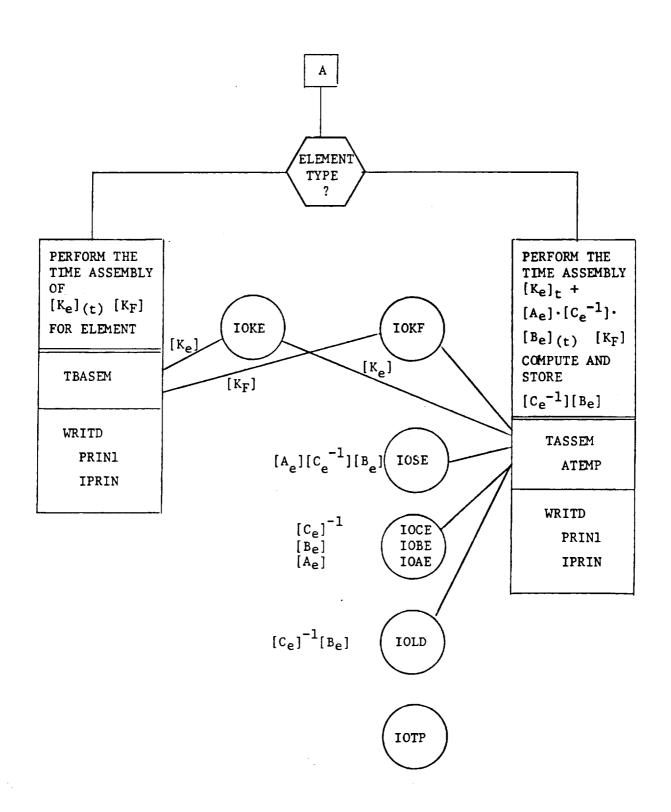
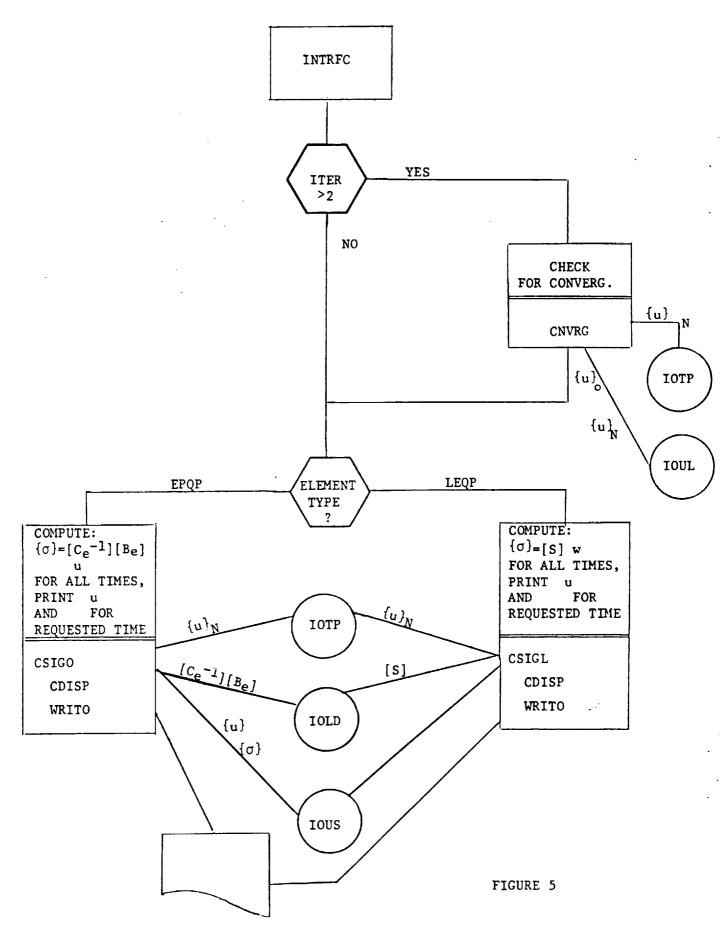


FIGURE 4 (Continued)



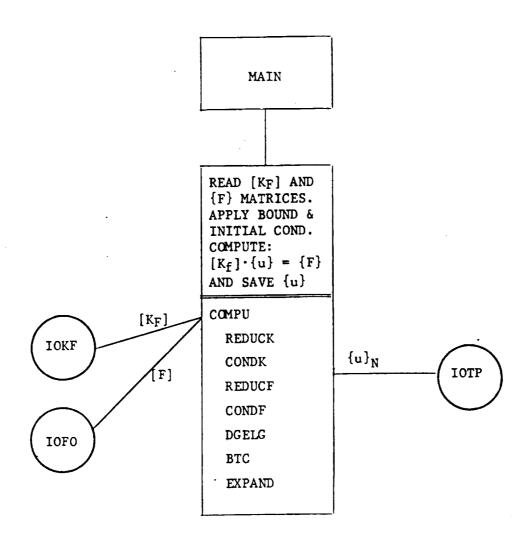


FIGURE 6

## 3.0 MATRIX DESCRIPTION AND LABELING

The basic matrices generated by the element computations consists of the  $[K_F]$ , [S], and  $\{F_F\}$  matrices. The matrices are stored on files "IOKF", "IOLD", and "IOFO" respectively. The basic matrices are illustrated in Figures 7 to 14 and are described as follows.

## 3.1 Stiffness [Kp] Matrices

Figure 7 depicts the time assembled  $[K_F]$  matrix for the EPQP element. Each numbered box represents a 168 by 168 partition and each partition is labeled from  $T_0$  to  $T_N$  (global time, were  $T_0$ =0.0). The numbering of the partitions signifies the order in which they are stored on the sequential file "IOKF". Similarly, Figure 8 depicts the time assembled  $[K_F]$  matrix for the LEQP element. It should be noted that rows I-84 of partition rows  $T_0$  and  $T_N$  are to be reduced from the matrix before using to solve for  $\{u\}$ . Finally, Figure 9 illustrates the labeling of each  $K_F$  partition.

## 3.1.1 [K<sub>F</sub>] MATRIX LABELING

The  $[K_F]$  matrix partitions are computed and time assembled  $(K_F = K_e + [A_e][C_e^{-1}][B_e])$ . The non-zero 168 by 168  $K_F$  partitions are stored on File IOKE. Figure 7 shows the  $K_F$  matrix-partition labeling and order of storing on file IOKF for the EPQP element. Similarly, Figure 8 represent the  $K_F$  matrix for the LEQP element. Figure 9 is a sketch of the labeling of each partition for both the EPQP and LEQP elements.

FIGURE 7: EPQP  $[K_{\mathbf{F}}]$  MATRIX PARTITIONS

where:  $T_{(L)} = input global times$ 

 $T_0 = 0.0$ 

Note:

- 1) Matrix partition col. no. 1 contains time  $\mathbf{T}_1$  partitions
- 2) Partitions are output to file IOKF in the order they are numbered in diagram
- 3) Each partition is a 168 by 168 matrix, see Figure 9 for labeling of partitions.

FIGURE 8. LEQP  $[K_{\overline{F}}]$  MATRIX PARTITIONS

	<sup>T</sup> 1	т2	т <sub>3</sub>	•	•	T <sub>N-1</sub>	T <sub>N</sub>	
T <sub>O</sub>	1	0	0	0	0	0	0	
T <sub>1</sub> .	2	4	0	0	0	-· o	0	
т2	3	5	7	0	0	o	o	
т <sub>3</sub>	0	6	8	10	0	0	0	
	0	0	9	•		0	0	
	0	0	0		•		0	
T <sub>N-1</sub>	0	0	0	0	•		//	=3(IC-1)+IF
T <sub>N</sub>	0	0	0	0	0	-		IC=COL. NO (1 to
7								IR=ROW NO. (1 to )

Where:  $T_{(i)} = Input global times$  $T_{O} = 0.0$ 

Note: 1) Partition matrix col. no. 1 contains time  $\mathbf{T}_1$  partitions

- 2) Partitions are output to file IOKF in the order they are numbered in diagram above
- 3) Each partition is a 168 by 168 matrix. See Figure 3 for labeling of partitions.

			x U GRID 1-28	x V GRID 1-28	x W GRID 1-28	х Ů GRID 1-28	x V GRID 1-28	x W GRID 1-28
	= F <sub>U</sub>	G 1 R I + D 28	<b>A</b>	•				
	= F <sub>V</sub>	G 1 R I + D 28						
T.	= F <sub>W</sub>	G 1 R I + D 28		[1	K <sub>F</sub> ] PARTI	IION		
T <sub>(i)</sub>	= F;	G 1 R I <sup>†</sup> D 28			168 BY	168		·
	= F:	G 1 R I + D 28						
	= F <sub>W</sub>	G 1 R I <sup>†</sup> D 28			-			

NOTE: GRID NOD ARE LOCAL

FIGURE 9

## 3.2 Stress [S] Matrices (EPQP Element)

The equation to compute the six (6) stress components at the 28 element grid points for time interval J is:

$$\{\sigma\}_{J} = [C_{\sigma}^{-1}][B_{o}^{1}]\{u_{o}\} +$$

$$J_{\Sigma} ([C_{\sigma}^{-1}] \cdot ([B_{o}^{1}] + [B_{T}^{(I-1)}]) \cdot \{u_{(I-1)}\}) +$$

$$I=2$$

$$[C_{\sigma}^{-1}][B_{T}^{J}]\{u_{(J)}\} -$$

$$J_{\Sigma} ([C_{\sigma}^{-1}]\{q_{(I)}\}) + \{\sigma_{o}\}$$

$$I=1$$

$$(1)$$

The time assembled resulting matrix equation, collecting like terms, could be represented as follows:

Figure 10 illustrates the  $[S_1]$  matrix which is computed in routine "TASSEM" and stored on file "IOLD" in 168 by 168 partitions. The number of each partition represents its location on the sequential output file "IOLD" as well as the terms of equation (1) above as listed below.

PARTITION TYPE CONTAINS

$$[C_{\sigma}^{-1}] \cdot [B_{\sigma}^{1}]$$
2, 4, 6, ... (Nx2) 
$$[C_{\sigma}^{-1}] \cdot [B_{T}^{IR}]; IR=1,N$$
3, 5, 7, ... (Nx2)-1 
$$[C_{\sigma}^{-1}] \cdot ([B_{\sigma}^{(IC+1)}] + [B_{T}^{IC}]); IC=1,(N-1)$$

Figure 11 illustrates the  $[S_2]$  matrix. Matrix  $[S_2]$  is simply the time assembled  $-[C^1]^{-1}$  matrix where  $[C^1]^{-1}[C_e]$  with the first column removed; i.e.,

$$[C_e] = \begin{bmatrix} -[C_o] & [C_o] & 0 & . & . & . & 0 & 0 \\ 0 & -[C_o] & [C_o] & . & . & . & . & . & . \\ 0 & -[C_o] & [C_o] & . & . & . & . & . & . & . \\ 0 & 0 & 0 & . & . & . & . & . & . & . \\ 0 & 0 & 0 & . & . & . & . & . & . \\ 0 & 0 & 0 & . & . & . & . & . & . \\ 0 & 0 & 0 & . & . & . & . & . & . \\ 0 & 0 & 0 & . & . & . & . & . & . \\ 0 & 0 & 0 & . & . & . & . & . & . \\ 0 & 0 & -[C_o] & [C_o] & 0 & . & . & 0 & 0 \\ -[C_o] & [C_o] & [C_o] & 0 & . & . & 0 & 0 \\ 0 & -[C_o] & [C_o] & [C_o] & . & . & 0 & 0 \\ 0 & 0 & -[C_o] & [C_o] & . & . & . & . & . \\ 0 & 0 & 0 & 0 & . & . & -[C_o] & [C_o] \end{bmatrix}$$

File "JOCE" contains a 168 x 168 partition,  $[C_o^{-1}]$ . The remaining

File "IOCE" contains a 168 x 168 partition,  $[C_0^{-1}]$ . The remaining variables namely,  $\{w\}$ ,  $\{q\}$ , and  $\{u_0/\sigma_0\}$  are stored on files "IOTP", "IOQT", and "IOUS" respectively.

Figure 12 contains a sketch of the partition labeling for both the  $[S_1]$  and  $[S_2]$  matrix partitions.

FIGURE 10. STRESS  $[S_1]$  MATRIX

		т <sub>о</sub>	т <sub>1</sub>	т <sub>2</sub>	т <sub>3</sub>		•	T <sub>N-1</sub>	T <sub>N</sub>	
	т <sub>1</sub>	1	2	0	0	•	•	0	0	
	т2	. 1	3	4	0	•	•	0	0	
	т <sub>3</sub>	1	3	. 5	6	•	•	0	0	
T <sub>(i)</sub>	•	•	•	•	•	•	•			(N)x(N+1) PARTITIONS
	•	•	ì	•	•	•	•		• •	
	T <sub>N-1</sub>	1	3	5	7	•	*•	N*2-2	0	
	T <sub>N</sub>	1	3	5	7	•	./	N*2-1	N*2	
	$= (IC-1):2+1 - \frac{IC}{(IR+1)}$									

IC = Column Number

IR = Row Number

T( ) = Input Global Times

 $T_0 = 0.0$ 

## NOTE:

- l) Matrix partition row number l contains time  $T_{f l}$  partitions
- 2) Partitions are output to file "IOLD" in the order they are numbered in diagram
- 3) Each partition is a  $168 \times 168$  matrix, see Figure 12 for labeling of partitions

FIGURE 11. STRESS [S2] MATRIX

		<sup>T</sup> 1	. T <sub>2</sub>	<b>T</b> 3	•	•	T <sub>N-1</sub>	TN	
	<b>T</b> 1	-[c <sub>σ</sub> -1]	0	0	•	•	0	0	
	т2	-[c <sub>o</sub> -1]	-[c <sub>o</sub> -1]	0	•	•	0	0	
	т <sub>3</sub>	-[c <sub>σ</sub> <sup>-1</sup> ]	-[c <sub>0</sub> -1]	-[c <sub>σ</sub> -1]		•	0	0	NXN
-[s <sub>2</sub> ] =	•	•	•	•	•	•	•	•	PARTITIONS
		•		•	•	•			
	$T_{N-1}$	-[c <sub>o</sub> -1]	-[c <sub>o</sub> -1]	-[c <sub>σ</sub> <sup>-1</sup> ]	•	•	-[c <sub>σ</sub> <sup>-1</sup> ]		
	T <sub>N</sub>	-[c <sub>o</sub> <sup>-1</sup> ]	-[c <sub>σ</sub> <sup>-1</sup> ]	-[c <sub>o</sub> <sup>-1</sup> ]	•	•	$-[c_{\sigma}^{-1}]$	-[c <sub>σ</sub> -1]	

## NOTE:

1) Each partition is identical and the 168 x 168 matrix  $[C_{\sigma}^{-1}]$  is stored on file "IOCE" by routine "EPQPA".

FIGURE 12. EPQP STRESS MATRIX PARTITION LABELING

				xU GRID 1-28	xV GRID 1-28	xW GRID 1-28	xU GRID 1-28	xV GRID 1-28	xW GRID 1-28
	= xx	G R I D	28			· · · ·			
	<b>-</b> yy	G R I D	1 28	· .	[S ]/	'IS 1 PART	ITION		
T	= ZZ	[S <sub>1</sub> ]/[S <sub>2</sub> ] PARTITION  R 168 X 168 D 28							
r(i)	= xy	G R I D	1 28		٠.				,
	= xz	G R I D	28						
	= yx	G R I D	1 28						<u>-</u> .

Note: Grid Numbers are Local

$$\sigma_{GRID,6,t}) = [S_1] \cdot \{u_t\} - [S_2] \{q_t\} + \begin{cases} \sigma_o \\ \sigma_o \end{cases}$$

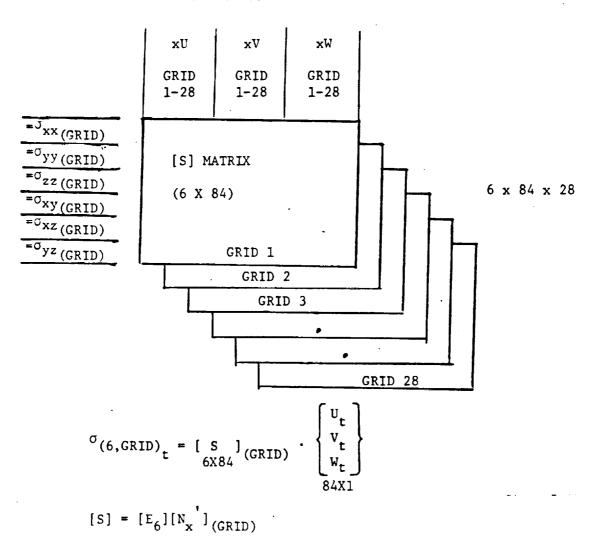
$$t=1-N \quad Nx \quad t=0-N \quad NxN \quad t=1-N \quad \sigma_o \\ (N+1) \quad Parti- \\ Partitions \quad tions \quad \sigma_o \end{cases}$$

## 3.3 Stress [S] Matrices (LEQP Element)

The 6 x 84 x 28 stress matrices for the LEQP element are generated in routine LEQP and stored on file "IOLD". The equation used to obtain [S] is  $[S]=[E_6][N_X^{-1}]_{(GRID)}$ , where  $[N_X^{-1}]$  is evaluated at each of the 28 grid points of the quadrilateral plate. Figure 13 illustrates the labeling of [S]. Since [S] for the LEQP element is time independent, the same matrix is used for each time increment in the problem solution and the velocity terms of the {u} vector are not used in this solution.

FIGURE 13. LEQP STRESS MATRIX LABELING





## 3.4 Element Force Vector {FF}

Figure 14 illustrates the force vector labeling and the time assembly of the element load vectors. The equations to adjust the force vectors for known initial conditions are as follows:

$$\{F_{F(o)}\} = \{F_o^{1}\} - [K_{oo}^{1}] \{u_o\} - [A_o^{1}] \{\sigma_o\}$$
(1)

$$\{F_{F}(1)\} = \{F_{O}^{2} + F_{T}^{1}\} - (\{K_{TO}^{1}\} + [A_{O}^{2}][C_{O}^{-1}][B_{O}^{1}]) \{u_{O}\} - ([A_{O}^{2}] + [A_{T}^{1}]) \{q_{O}\} + [A_{O}^{2}][C_{O}^{-1}] \{q_{I}\}$$

If N=1 
$$\{F_{F(1)}\} = \{F_{F(1)}\} - [A_{T}^{1}] [C_{\sigma}^{-1}] [B_{\sigma}^{1}] \{u_{\sigma}\}$$

$$\{F_{F(J)}\} = \{F_{\sigma}^{(J+1)} + F_{T}^{J}\} - ([A_{\sigma}^{(J+1)}] + [A_{T}^{J}]) [C_{\sigma}^{-1}] [B_{\sigma}^{1}] \{u_{\sigma}\} - J^{-1} (([A_{\sigma}^{(J+1)}] + [A_{T}^{J}]) \{\sigma_{\sigma}\} + J^{-1} (([A_{\sigma}^{(J+1)}] + [A_{T}^{(J)}]) [C_{\sigma}^{-1}] \{q_{L}\} + L=1$$

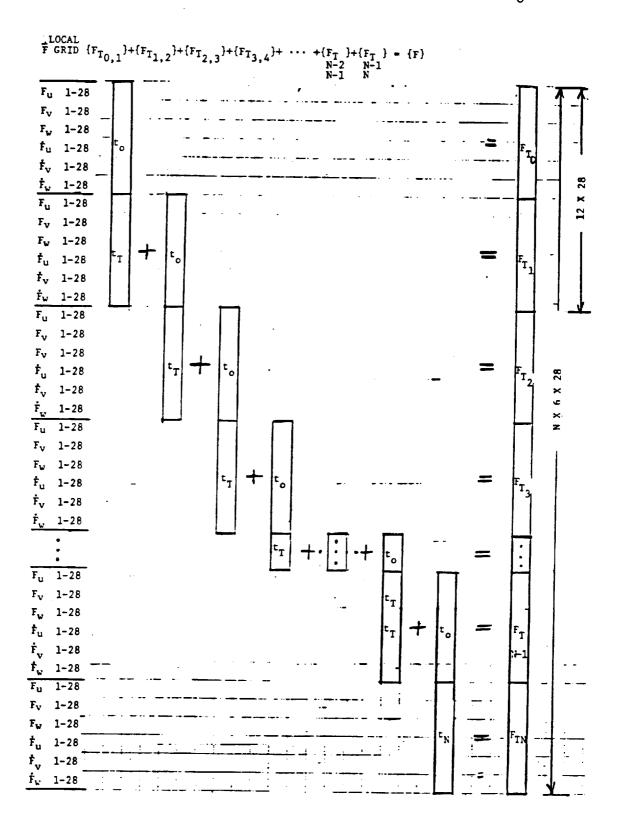
 $[A_0^{(J+1)}] [C_0^{-1}] \{q_1\}$ 

The first force vector is written to file  $IOFF^{(1)}$  for solution of the problem  $[K_F]$   $\{u\}$  =  $\{F_F\}$ .

NOTE:

(1) The above computations are not currently performed in the program. It is assumed that  $\{u_0\} = \{\sigma_0\} = \{q\} = \{0\}$  and therefore file "IOFO" contains the  $\{F_{OIT}^{\ L}\}$  vectors (unassembled). Routine "COMPU" currently preferring the time assembled.

FIGURE 14. {F} VECTOR TIME ASSEMBLY AND LABELING N=NTIME,  $T_0$ =0.0



#### 4.0 EXTERNAL DATA SET STRUCTURE

The SFENES program uses 18 data sets during execution. The dataset variable unit names are stored in labeled common "IO" and are initialized to the integer unit number in routine "INTRFC". Table 1 contains the dataset names, default unit number, routines using the dataset and a brief description of the contents of each dataset. Table 2 lists the record descriptions and record formats contained in each dataset. All datasets contain binary output except "IOIN" and "I6". The binary datasets in several instances contain variable length records.

CONTENTS	Element load codes and the computed or input force tables. This data is destroyed after total element load vectors are computed.	Element stress matrix (LEQP element)	Element matrix computation $[C_{e}]^{-1} \cdot [B_{e}]$ - $(EPQP \ Stress \ Computation) \ results$	The element input data (name, local coordinates, material properties, load codes)	The element stress component values, displacement and velocity values at each time increment and grid point [S], [U]	User element input data - load, material prop., titles, times, initial conditions	Not Used	Printed output file (133 character records)
IN/OUT	OUT IN IN	OUT IN	. our	OUT IN	OUT IN IN/OUT IN/OUT	NI NI		00 00 00 00 00 00 00 00
ROUTINE USING	INPUT EPQPA RDOTDT	LEQP	ATEMP	INPUT	INPUT EPQP19 CSIGL CSIGO	INPUT		CSIGL CSIGO ERROR INPUT IPRIN PRINI
UNIT NO.	1 (Binary)			2 (Binary)	3 (Binary)	5 (Formatted)	4	(Formatted)
UNIT	IOLD			IOEL	IOUS	IOIN	1001	91

TABLE 1 (CONTINUED)

CONTENTS	Spatial computations and integration results $[B_{28}]^{-1}$ , $[D] \cdot [N_{\mathbf{x}}1]$ , $[P_{28}] \cdot [B_{28}]^{-1}$ , $[P_{28}] \cdot [M]$ , $\int_{\mathbf{v}} [N_{\mathbf{x}}1] \cdot [N_{\mathbf{x}}1]  d\mathbf{v}$ , $\int_{\mathbf{s}\mathbf{v}} ([N_{\mathbf{x}}^{\mathrm{B}}]^{\mathrm{T}} - [N_{\mathbf{x}}]^{\mathrm{T}}) \cdot [\mathbf{I}] \cdot [N_{\mathbf{x}}^{\mathrm{T}}]  d\mathbf{s}$ , $\int_{\mathbf{v}} [N]^{\mathrm{T}} \cdot [\mathbf{N}^{\mathrm{X}}]^{\mathrm{T}} - [N_{\mathbf{x}}]^{\mathrm{T}} \cdot [N_{\mathbf{y}}]^{\mathrm{T}} - [N_{\mathbf{y}}]^{\mathrm{T}} \cdot [N_{\mathbf{j}}]^{\mathrm{T}} - [N_{\mathbf{j}}]^{\mathrm{T}} \cdot [N_{\mathbf{j}}]^{\mathrm{T}} \cdot [N_{\mathbf{j}}]^{\mathrm{T}} - [N_{\mathbf{j}}]^{\mathrm{T}} \cdot [N_{$	Temporary work file containing a copy of IOPR $Temporary \ \ work \ file \ containing \ a \ copy \ of \ \{A_e\} \ \ matrix$	Temporary work file containing the computed {u} vector for all major time increments	Contains matrix $[C_e]^{-1}$	Element [K <sub>e</sub> ] matrix for each time increment
IN/OUT	OUT IN IN	OUT IN/OUT IN/OUT IN/OUT	TOO NI NI	OUT	OUT NI OUT NI
ROUTINE USING	ЕРОРА ЕРОРВ LEQP	EPQPA EPQPB LEQP ATEMP	COMPU CSIGL CSIGO	EPQPA ATEMP	EPQPB TASSEM LEQP TBASEM
UNIT NO.	7 (Binary)	8 (Binary)		9 (Binary)	10 (Binary)
UNIT	IOPR	IOTP		IOCE	IOKE

TABLE 1 (CONTINUED)

CONTENTS	Element [A <sub>e</sub> ] matrix partitions for each increment	Element [B <sub>e</sub> ] matrix partitions for each increment	Contains coefficient arrays used to compute the [E7] matrices for each element time increment. One contains previous, the other current coefficients.	Integrated element force data for each grid and time point {F}	Contains the temporaly assembled [KF] matrix	Temporary storage of the $[A_{f e}]$ $[C_{f e}^{-1}]$ $[B_{f e}]$ computation	The resulting displacement and velocities from the previous iterations solution. Saved for comparison to the results of the next iteration.
IN/OUT	OUT	OUT IN	OUT NI	DUT IN	OUT OUT IN	OUT	IN/OUT
ROUTINE	EPQPB ATEMP	ЕРОРВ АТЕМР	QCOEF EPQP19	LOADD COMPU	TASSEM TBASEM COMPU	ATEMP TASSEM	CNVRG
UNIT NO.	11 (Binary)	12 (Binary)	13 14 (Binary)	15 (Binary)	16 (Binary)	17 (Binary)	18 (Binary)
UNIT	IOAE	IOBE	10A1 10A0	IOFO	IOKF	IOSE	IOUL

FILE NAME	RECORD DESCRIPTION	RECORD FORMAT
IOLD (Load Data)	l Element Load Code	((LCODE(I,J),I=1,7),J=1,6) 42 integers/record
	2 Element forces at NGRID affected grids for this load type and for D.O.F. u,v,w	((FVO(I,J),J=1,NGRID),I=1,3) NGRID = 10, 12, or 28 .: 30, 36 or 84 real data/record
	If ID=2 only element rate of force at NGRID affected grids for this lead type and D.O.F. u,v,w -	((FVT(I,J),J=1,NGRID),I=1,3) NGRID = 10, 12 or 28 :30, 36, or 84 real data/record
	NOTE: Records (2) or (2 and load type and for each	3) are repeated for each input h time point.
	The data is overwritt	en by:
(LEQP Stress Matrix)	1 Element stress matrix 6 components, computed from 84 displacement D.O.F. for all 28 grids	(((STRM(I,J,K),I=1,6),J=1,84), K=1,28) 14112 real data/record
	or by:	
(EPQP Stress Matrices [C <sub>e</sub> ]-1·[B <sub>e</sub> ]	1 Element stress matrix partition [C ] <sup>-1</sup> ([B <sub>O</sub> ] <sup>N+</sup> [B <sub>T</sub> ] <sup>N-1</sup> )	((DOT(I,J),I=1,168),J=1,168) 28224 real data/record
	2 Element stress matrix partition [C ] $^{-1} \cdot [B_T]^N$	((DT(I,J),I=1,168),J=1,168) 28284 real data/record
	NOTE: Records 1 and 2 are r (NTIME-1))	epeated (NTIME-1) times (N=1,
IOEL (Element Input Data)	1 Element name and code  2 Element local coord., material prop., and load codes	ANAM, NELEM 21 real and 1 integer/record  (X(I),Z(I)=1,28), (H(J),J=1,4), RHO, AMU, ALPHA, NU, E, YSTR, AJ, (ILOAD(T),I=1,7), NONLIN 94 real and 9 integer data items/ record

FILE NAME	RECORD DESCRIPTION	RECORD FORMAT
IOUS (Element Input Data)	l Element stress components (6 each) for 28 grid poin points	((SIG(I,J), I=1,28), J=1,6) 168 real data/record
	2 Element disp-vel. compo- nents (u,v,w,ů,v,ů) for 28 grid points	((U1(I,J),I=1,28),J=1,6) 168 real data/record
	NOTE: Records 1 and 2 are	repeated NTIME times
IOIN (Element Input Formatted)	Several	See User Input Description
IOPR (Spatial	I The $[B_{28}]^{-1}$ matrix	((B28(I,J),I=1,28),J=1,28) 784 real data/record
Computation Data)	2 The [D] $[N_X']$ computation for all 28 grid points	(((DNXPP(I,J,K),I=1,7),J=1,84), K=1,28) 16414 real data/record
	3 The $\{P_{28}\} \cdot [B_{28}]^{-1}$ computation for 28 grids	((SIGNX(I,J),I=1,28),J=1,28) 784 real data/record I=polynomial terms 1-28 J=grids 1-28
	4 The evaluation of $\{P_{28}\}$ for 28 grids	((P28(I,J),I=1,28), J+1,28) 784 real data/record I=polynomial terms 1-28 J=grids 1-28
	5 The mass computation $[B_{28}^{-1}]^{T} \cdot [\int_{V} P_{28}^{T} \cdot P_{28}^{T} dV] \cdot [B_{28}^{-1}]$	((CUUM(I,J),I=1,28),J=1,28) 784 real data/record
	6 The [K] computation [B <sub>28</sub> -1] <sup>T</sup> ·[f <sub>V</sub> P <sub>28</sub> (L)· P <sub>I</sub> <sup>T</sup> ·P <sub>j</sub> dV]·[B <sub>28</sub> -1]	(((CIJK(M,N,K,),M=1,28),N=1,28) K=1,9) 7056 real data/record
	NOTE: Record 6 is repeated for LEQP NLIN=1 for EPQP NLIN=28	NLIN TIMES and L=1, NLIN:

#### FILE NAME

#### RECORD DESCRIPTION

#### RECORD FORMAT

IOPR (Cont.)

(((CIJXE(M,N,K),M=1,28),N=1,28),4704 real data/record 

NOTE: Record is repeated NLIN Times and L=1, NLIN; For LEQP, NLIN=1 For EPQP, NLIN=28

Record: 2\*NLIN+6 The [A<sub>e</sub>] spatial computations - EPQP only  $\begin{cases} ([B_{28}^{-1}]^{T} \cdot [[P_{1}^{T} \cdot P_{28} dV]] + \\ \sum_{J=1}^{4} ((B_{10}^{-1})^{J})^{T} \cdot [[P_{1}^{T} \cdot P_{28} dV]] + \\ P_{28(J)}^{dS} - [B_{28}^{-1}]^{T} \cdot \\ [[P_{28}^{T} \cdot P_{28}]^{T} \cdot P_{28}(J)^{dS}] \cdot \end{cases}$   $\begin{cases} K = 1 & 2 & 3 \\ I = x & y & z \\ ANG(K, J) = SIN & COS & 0.0 \end{cases}$  $\langle ANG(K,J)\rangle \cdot [B_{2R}^{-1}]$ 

(((CIJA(M,N,K),M=1,28),N=1,28),K=1,3)2352 real data/record

Record: 2\*NLIN+7 The  $[B_e]$  spatial computations - EPQP only  $[B_{28}^{-1}]^{T} \cdot [\sqrt{P_{28}(L) \cdot P_{28}^{T}}]$  $[P_{1}^{dV}] \cdot [B_{28}^{-1}]$ 

Note: Record is repeated NLIN times and L=1, NLIN; For LEQP NLIN=1 For EPQP NLIN=28

FILE NAME	RECORD DESCRIPTION	RECORD FORMAT					
IOTP	Same as File IOPR						
Temporary Storage)	- or -						
Storage)		((ABIN(I,J),I=1,168),J=1,168)					
	Record 1 - EPQP only $[A_0]^N$ partition of $[A_e]$ matrix	28224 real data/record					
	Record 2 - $[A_T]^N$ partition of $[A_e]$ matrix	((ABIN(I,J),I=1,168),J=1,168) 28224 real data/record					
	NOTE: Records 1 and 2 are repeated NTIME times						
-	- or -						
	Record 1						
	Displacement Vector {u}	(U1(K,I),K=1,168); 168 real data/record					
	NOTE: Record is repeated for	or I=2, NTIME					
IOCE ([Ce-1] Matrix)	Record 1 - EPQP Only [Ce <sup>-1</sup> ] matrix	((CE(I,J),I=1,168),J=1,168 28224 real data/record					
IOKE ([K <sub>e</sub> ] Matrix)	Record 1 [K <sub>e</sub> ] matrix partition	((AKE(I,T),I-1,336),J=1,336) 112896 real data/record					
	NOTE: Record is repeated (1	NTIME-1) times					
IOAE ([Ae] Matrix)	Record 1 - EPQP Only $[A_O]^N$ partition of $[A_e]$ matrix	((AE(I,J),I=1,168),J=1,168) 28224 real data/record					
	Record 2 $[A_T]^N$ partition of $[A_e]$	((AE(I,J),I=169,336),J=1,168) 28224 real data/record					
	NOTE: Records 1 and 2 are 1	repeated (NTIME-1) times					
IOBE ([Be] Matrix)	Record 1 - EPQP Only $\begin{bmatrix} B_0 \end{bmatrix}^N$ partition of $\begin{bmatrix} B_e \end{bmatrix}$	((BE1(I,J),I=1,168),J=1,168)					
	Record 2 $[B_7]^N$ partition of $[B_e]$	((BE2(I,J),I=1,168),J-1,168)					
	NOTE: Records 1 and 2 are	repeated (NTIME-1) times					

FILE NAME	RECORD DESCRIPTION	RECORD FORMAT	
IOA1 <u>IOAO</u> (E <sub>7</sub> Coeff Arrays)	Record 1 - EPQP Only Arrays AC, AT1, AT2, AT3, AT4, Al1 (7, 7, 28, 4)	(AC, AT1, AT2, AT3, AT4) All dimensioned (7, 7, 28, 4) 27440 real data/record	
	NOTE: Record is repeated (NTIMES-1) times		
IOFO (Integ. Force Vector)	Record 1 Vector {F}	(((VLOAD(I,J,K),I=1,28),J=1,3), K-1,4) 336 real data/record I=GRID, J=D.O.F. (u,v,w,) K=0, 0, T, T	
	NOTE: Record is repeated (	Record is repeated (NTIME-1) times	
IOKF ([K <sub>F</sub> ] Matrix)	Record 1 Partition (M,N) of the [K <sub>r</sub> ] matrix	((ACB(I,J),I=1,168),J=1,168) 28224 real data record	
	NOTE: Record repeated ( $\Sigma$	(NTIME - (I-1))-1 times	
$\frac{\text{IOSE}}{([A_e] \cdot [C_e]^{-1}} \cdot [B_e])$	Record 1 - EPQP Only Partition (M,N) of [A <sub>e</sub> ]·[C <sub>e</sub> <sup>-1</sup> ]·[B <sub>e</sub> ]	((ACB(I,J),I=1,168)J=1,168) 28224 real data/record	
	NOTE: Record repeated ( $\Sigma$	(NTIME-I))-l times	
10UL (u, v, w, ů, v, w)	Record 1 Displacement/velcoity vector from the previous iteration {u}	(UNEW(I,J),I=1,168	
,	NOTE: Record is repeated for J=2, NTIME		

# SLAVE FINITE ELEMENT FOR

# NONLINEAR ANALYSIS OF ENGINE STRUCTURES

#### USER'S MANUAL

#### I.O INTRODUCTION

The purpose of the SFENES program is to provide the user with validation of the basic matrices needed to compute the transient displacements of a group of non-linear elements and to provide for element stress recovery unpon convergence of the problem. In addition, the element matrices routines have been provided to compute resultant displacements, checks for convergence, output displacement and stress and perform element stress recovery. The program is limited by the following factors:

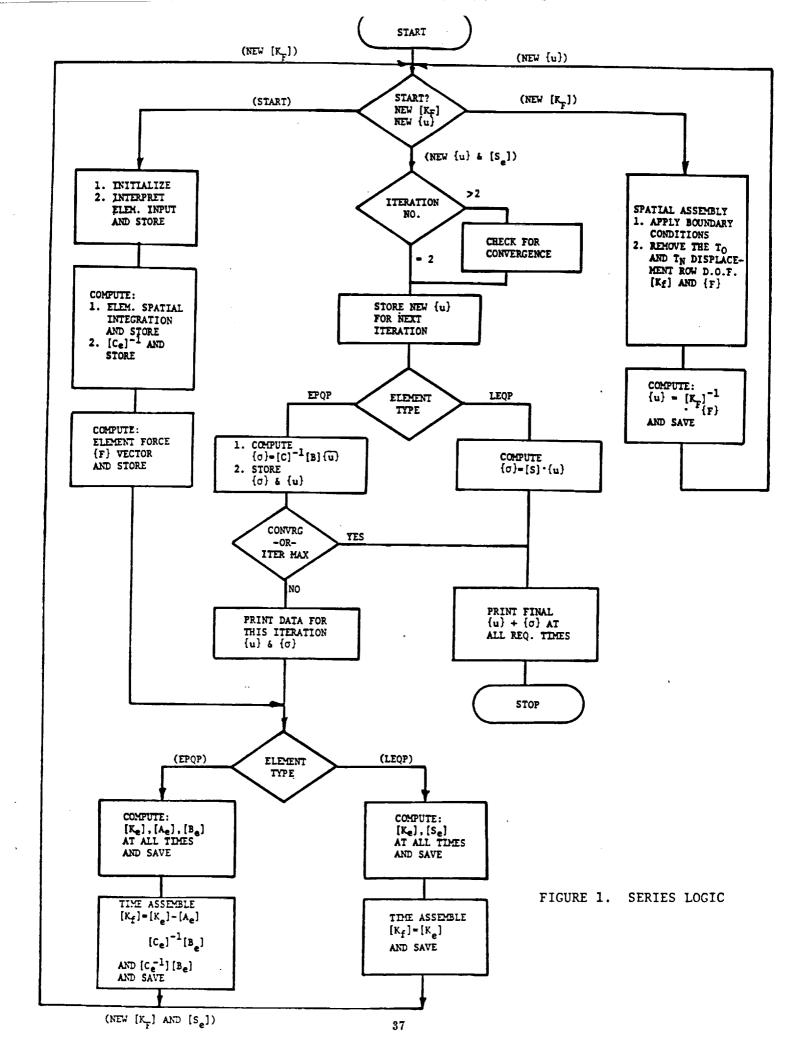
- The only element currently provided is the quadrilateral plate element with either linear elastic analysis or non-linear elastic-plastic analysis.
- Only a single structural element may be analyzed (multiple time intervals).
- 3. All input and output is in terms of the "local" coordinate system.
- 4. A maximum of 25 time points (solution times) may be requested. Output displacement times are unlimited.

The following pages describe the program and its input/output features.

# 2.0 SFENES OVERVIEW

The Slave Finite Element procedure is described here. Reference to Figure 1 will aid in the description of the solution algorithm. The element input data is interpreted and stored for later reference. All of the required spatial definite integrals are computed and stored as is the one time computation of the [C]<sup>-1</sup> matrix. The input element lead data (if any) is combined and integrated over the time intervals. The force vector {F} is then stored for later use in computation of {u}.

The iterative procedure starts with the computation of the matrices  $[K_e]$  and either  $[A_e]$ ,  $[B_e]$  for the EPQP case or  $[S_e]$  for the LEQP case. For the EPQP case the initial iteration assumes that the structural element is elastic for all time intervals but for each subsequent iteration the elastic and/or plastic regions are computed. The structural matrices are then assembled for global time. For the EPQP case final time assembled  $[K_{\mbox{\it F}}]$  matrix is computed as  $[K_F] = [K_e] = [A_c] [C_e]^{-1} [B_e]$  and the matrix multiplication  $\left[\mathrm{C_{e}}\right]^{-1}$   $\left[\mathrm{B_{e}}\right]$  is stored for later stress computation. The linear LEQP case simply involves the time assembly of each of the  $[K_e]$  matrices as computed over the specified local time intervals. The [ $K_F$ ] matrix and the  $\{F\}$  vectors are then reduced based upon the input boundary conditions as well as the removal of the tows corresponding to displacements at times  $T_0$  and  $T_N$  (rows 1-84) and rows ((N\*168)+1 to (N\*168)+84) where N = the number of input time points minus 1 (1<N<24). The reduced  $[K_F]$  matrix and  $\{F\}$  vector are used to compute the new



{u} vector ( $[K_F]$  {u} = {F}). The updated value of {u} is checked for convergence and then stored for the next iteration.

The vector  $\{u\}$  and the stress matrices  $[C_e]^{-1}$   $[B_e]$  or  $[S_e]$  are used to compute the stress vector  $\{\sigma\} = [C_e]^{-1}$   $[Be_e]$   $\{u\}$  or  $\{\sigma\} = [S_e]$   $\{u\}$ . The displacement and stress vectors are then stored for use in computation of the  $[E_7]$  matrices for the next iteration. After each iteration the displacement and stresses are printed for the input time intervals only. When the problem converges the displacements are also printed at the requested output interval. The iterative procedure continues until convergence or maximum iteration is achieved.

# 3.0 QUADRILATERAL PLATE DESCRIPTION

One finite element is provided in SFENES and that is the quadrilateral plate element. Figure 2 depicts the local coordinate grid labeling and the orientation requirements. The analysis procedure permits transient analysis of the plate as either a linear elastic element (LEQP) or as a non-linear elastic-plastic element (EPQP).

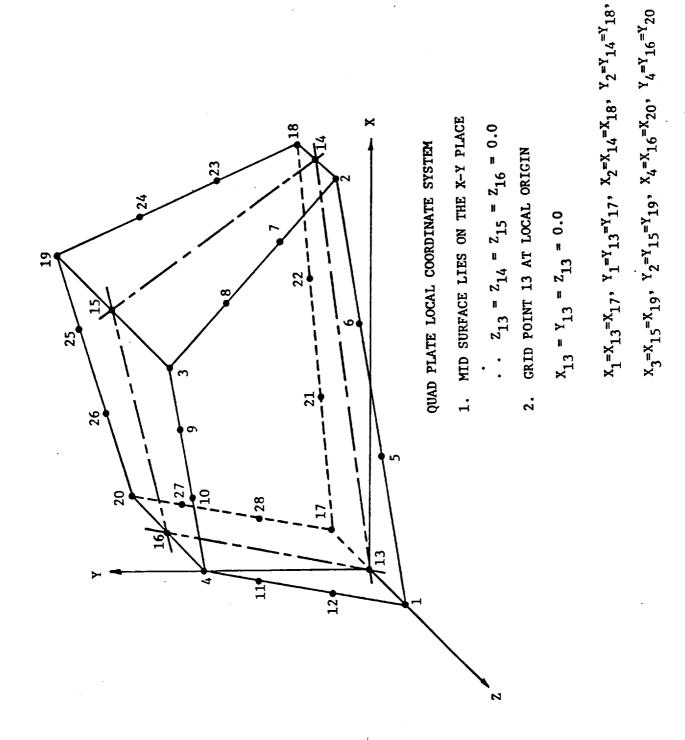


FIGURE 2. QUADRILATERAL PLATE LOCAL GRID IDENTIFICATION

# 4.0 INPUT OVERVIEW

The input requirements and options are described below. The following general rules must be observed.

- Each input line consists of 8 fields of 10 characters each. In some cases a field is subdivided for integer or alpha input.
- 2. Integer data and floating point "E" format data must be right justified in their respective input fields.
- 3. One of the material prop inputs is required.

Listed below is a brief summary of the input options and notes regarding order and required usage.

TITLE Title information

TIME (Required and must preceed and load input)

The number of time increments +1 and the values of beginning/end of the time increments.

EPQP Contains material properties and local grid point coordinates for the linear plastic quad plate element.

LEQP Contains material properties and local grid point coordinates for the linear elastic quad plate element.

LOAD Defines element body force, pressure and traction load (must be preceded byTIME input option).

ISTRES Defines any initial stress component values (at TIME = 0.0)

ISISP Defines any initial displacement and velocity component
values (at TIME = 0.0)

PDOF Displacement and STRESS output print selection (Default = Print All D.O.F.)

PGRD	Local grid number output print control (Default = Print
	All Grid Points).
ICND	Boundary conditions, grid identifications (Default=
	None Bounded Out)
CONT	Execution control parameters - maximum iterations,
	convergence criteria and output format.
END	Require last input for the element.

# 5.0 INPUT DESCRIPTION

Figure 3 contains an illustrative example of typical user input. A detailed description of each line and field of input, that may be entered for each of the input options summarized in Section 4.0, is contained on the following pages. Input record description contained on these pages includes:

- 1) Input Code, i.e., "TIME"
- 2) Record Format, i.e., Field Usage and Mnemonic
- 3) Record Example, as contained on Figure 3
- 4) Field Description verbal explanation of each field in the record along with data type, limits and defaults if any.
- 5) Qualifying Notes

  All input is formatted and is currently read from logical unit 5

  "IOIN".

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·	03/ 02/	84 -												
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)F	C40.0 C20.0 C10.0	1 1 1	.0	1	1 1 1	1 1 1	1	1 1 1	1	1	1	i 1	1 1	1, 1
)F	C40.0 C20.0 C10.0	0.	.0	0.	1	1 1	1	1	1	0	•0	1 1 1 18	1 1 20	1. 1

FIGURE 3. INPUT SAMPLE

T(1)   (1)	T(4)
-	

TIME (REQUIRED)

INPUT

FIELD DESCRIPTION:

FIELD

CARD

CONTENTS	THE ALPHANUMERICS "TIME" THE NUMBER OF TIME VALUES WHICH FOLLOW (INTEGER 2 <nt<25) (real="" 6="" data="" first="" the="" time="" values="">0.0) THE 7TH THRU THE 25TH TIME VALUES (REAL DATA &gt;0.0)</nt<25)>
FIELD NO.	N C4 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
FIELD	 TIME NT T(1)-T(6) T(7)-T(25)
C)	4 1 4

FORMAT:	
TITLE NC	
BO CHARACTERS OF ALPHANUMERIC DATA	 
EXAMPLE:	

TITLE

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CONTENTS	THE ALPHANUMERIC "TITLE" THE NUMBER OF TITLE CARDS THAT FOLLOW (INTEGER, 1 <nc<3) ALPHANUMERICS WHICH ARE TO IDENTIFY OUTPUT LISTINGS</nc<3) 
FIELD NO.	1 1 1 1 8
FIELD	TITLE NC
CARD NO.	4 4

**c**<sup>C</sup> · .

#### FORMAT:

	'				·			
LEPOP N	IL!		E .	G	AMU	RHO	ALFHA :	
:	1 Y	STR :	NU			•		
: CORD I	GIX	ENTER :	YENTER :	ZENTER	H			-

#### EXAMPLE:

EFQP	Ŏ1		•	11.1538E+7	0.3	.0002588	· - • •	i
i		3000.	12.0	į		;	!	
CORD	13	0.0	•	10.0	1.0	;	: !	
CORD	16	0.0	•	10.0	11.0	;	! !	
ICGRD	14	10.0		10.0	11.0	:	!	
CORD		10.0	•	•	1.0	;	;	;

#### FIELD DESCRIPTION:

#### \_\_\_\_\_

CARD	FIELD	FIELD	CONTENTS
NO.	LABEL	NO.	
1	EF:QP	1	THE ALPHANUMERIC "EPQP"
1	NL	1	CODE TO FERFORM LINEAR OR NONLINEAR (ITERATIVE) COMPUTATIONS;
			(INTEGER, 00 = LINEAR, 01 = NUNLINEAR)
1	E	3	YOUNG'S MODULUS (REAL > 0.0)
1	G	4	SHEAR MODULUS (REAL > 0.0)
1	AMU	5	POISSON'S RATIO (REAL, 0.0KAMUK0.5)
1	RHO	6	MASS DENSITY (REAL)
1	ALPHA	7	MATERIAL PARAMETER IN PLASTICITY LAW (REAL)
1	AJ	8	MATERIAL FARAMETER IN PLASTICITY LAW (REAL)
2	YSTR	2	NOMINAL YIELD STRESS (REAL)
2	NU	3	ALTERNATE POISSON'S RATIO (NOT USED) (REAL)
3 - 6	CORD	1	THE ALFHANUMERIC "CORD"
2 - 6	IG	1	THE LOCAL GRID NUMBER (INTEGER, 13<1G<16)
3 - 6	XENTER	2	THE LOCAL X-COORDINATE OF GRID POINT IG (REAL)
3 - 6	YENTER	3	THE LOCAL Y-COGRDINATE OF GRID POINT IG (REAL)
3 - 6	ZENTER	4	THE LOCAL Z-COORDINATE OF GRID POINT IG (REAL)
3 - 6	Η ,	5	THE THICKNESS OF PLATE AT CORNER CONTAINING LOCAL GRID FOINT IG (REAL>O.)

NOTE: 4 CORD CARDS ARE REQUIRED, ONE EACH FOR LOCAL GRID FOINTS 13,14,15 AND 16. IN THE LOCAL COORDINATE SYSTEM GRID POINT 13 MUST BE AT THE ORIGIN AND POINTS 13 - 16 MUST ALL LIE IN THE X-Y PLANE.

PLATE ELEMENT
QUADRILATERAL
ELASTIC
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FORMAT:

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				YENTER	
				IG! XENTER	
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-			-	CORD	-

EXAMPLE:

LECIP	LEGP			10.3	12.58BE-4	
CORD	CORD 13:0.0	0.0	0.0	1.0		
CORD	CORD 16:0.0	10.0	0.0	11.0		
CORD	CORD 14:10.0		0.0	11.0		
CORD	CORD 15110.0 10.0	110.0	10.0	11.0		

FIELD DESCRIPTION:

CONTENTS		
FIELD	NO.	
FIELD	LABEL	
CARD	<b>.</b>	

	(REAL >0.)
	16
	GRID POINT
THE ALPHANUMERICS "LEGP" SHEAR MODULUS (REAL > 0.0) POISSON'S RATIO (REAL, 0.0   POISSON'S RATIO (REAL, 0.0   MASS DENSITY (REAL > 0.0)   THE ALPHANUMERIC "CORD"   THE LOCAL GRID NUMBER (INTEGER, 13   THE LOCAL X-COORDINATE OF GRID POINT 16 (REAL)   THE LOCAL Y-COORDINATE OF GRID POINT 16 (REAL)   THE LOCAL Z-COORDINATE OF GRID POINT 16 (REAL)	CEAL SOLUTION LUCAL GRID POINT IG (REAL >0.)
<b>ተ404++2044</b>	1
1 LEQP 1 6 1 ANU 1 RHO 2 - 5 CORD 2 - 5 IG 2 - 5 XENTER 2 - 5 ZENTER 2 - 5 ZENTER	NOT CONTRACT OF CO

4 CORD CARDS ARE REQUIRED, ONE EACH FOR LOCAL GRID POINTS 13,14,15 AND 16. IN THE LOCAL COORDINATE SYSTEM GRID POINT 13 MUST BE AT THE ORIGIN AND POINTS 13 - 16 MUST ALL LIE IN THE X-Y PLANF. NOTE:

48

INPUT - LOAD - ELEMENT LOAD DATA

FORMAT: FOR EQUATION TYPE INPUT It=1 OR It=2

	XWYWZWTW	XMYWZWTW
	 	!!!
1	XuYuZ	χαγα
Itidipin	COEFFu	COEFFu
LOAD II		

FORMAT: FOR TABLE TYPE INPUT It=3

·		Fdt(IG,w)	!
	.]	Fdt (IG,v) 11	
 	· · · · · · · · · · · · · · · · · · ·	:Fdt(IG,u) :Fdt(IG,v)	
 		IF(IG,w)	
		F(IG, v)	
 Ilitidipinis		161F(16,u)	
ורטאם			

# EXAMPLE:

	- !		. <u>-</u>		•
				. 1000	. •
				10000	•
				-10000.0	
				100000000	
				0.0	
				1000000000	
-		02:02010101011		0110.0	
		05		0	
		ILOAD			

CONTENTS	THE ALPHANUMERICS "LOAD"  LOAD TYFE CODE (INTEGER; 1, 2 OR 3)  1 = BODY FORCE LOAD  2 = FRESSURE LOAD		3 = FORCES INPUT DIRECTLY AS TABLES CODE INDICATING IF ONLY FORCE OR FORCE AND RATE OF FORCE INPUT FOLLOWS (INTEGER, 1 OR 2)	<pre>1 = FORCE EQUATIONS OR TABLES ONLY 2 = FORCE AND RATE OF FORCE EQUATIONS OR TABLES CODE DESCRIBING TIME FUNCTION COMPUTATIONS (INTEGER, 1 OR 2) 1 = NON-HERMITIAN COMPUTATIONS</pre>	2 = HERMITIAN TIME FUNCTIONS (Id MUST BE 2) (In)*(Id) EQUALS THE NUMBER OF INPUT LINES THAT FOLLOW FOR EQUATION TYPE INPUT AND (In)*(NUMBER OF TIME POINTS) EQUALS THE NUMBER OF INPUT LINES THAT FOLLOW FOR TABLE TYPE INPUT, PERMISSIBLE VALUES OF IN ARE GIVEN IN THE TABLE BELOW.
FIELD NO.	<b>ન</b> ન	и	и	И	<b>14</b>
FIELD LABEL	LUAD '	I t	Id	Ip	r I
CARD NO.		Ţ	1	<b></b>	<b></b> 50

VALUES OF In IF:

10	1 - 10	10,4	
		12,4	
	N	28	
	1,2		
	N	1,2 1 - 28 1 - 12 1 -	1,2 1 - 28 1 - 12 1 - 3 3 28,8 12,4 10,

CODE DESCRIBING THE SURFACE OR EDGE THAT PRESSURE OR TRACTION LOAD IS APPLIED ON ENTER O FOR BODY FORCE TYPE LOAD. OTHERWISE, REFER TO TABLE BELOW.

N

IURE LOAD SURFACE TRACTION LOAD EDGE RID NUMBERS GRID NUMBERS
SURFACE ERS
E LOAD SUR
PRESSURE LOAD GRID NUMBE
<u>មា</u>

1,2,17,18 EDGF	2.3.28.19 FIGE	M. 4. 19 20 Chen	4.1.20.17 FDGE
1 TO 12 (TOP)	1/ 10 28 (BOTTOM)	Z.D.	Z.S.

HNMA

NWA

FIELD DESCRIPTION:LINE 2 TO (In\*Id+1) FOR EQUATION TYPE INPUT

COEFFICIENTS AND EXPONENTS FOR RATE OF FORCE COMPUTATIONS SIMILAR TO THOSE DESCRIBED ABOVE. (Iu\*Id+1)

EXPONENTS OF THE X,Y,Z,T VARIABLES RESPECTIVELY, OF THIS TERM FOR COMPUTATION OF  $\bf w$  DIRECTION FORCE (INTEGERS  $> \phi$ )

COEFFICIENT OF THIS TERM FOR COMPUTATION OF w DIRECTION FORCE (REAL)

OF v DIRECTION FORCE (INTEGERS > 0)

9 7

2 -In+1CBEFFw 2 -In+1Xw,Yw,Zw,Tw

In+2

2

FIELD DESCRIPTION:LINE 2 TO (In\*NTIME+1) FOR TABLE INFUT

	(REAL)
	THIS FORCE IS APPLIED (INTEGER) FORCE (REAL) FORCE (REAL) RATE OF FORCE (BLANK IF Id = 1) RATE OF FORCE (BLANK IF Id = 1) RATE OF FORCE (BLANK IF Id = 1)
CONTENTS	GRID NUMBER AT WHICH U-COMPONENT VALUE OF W-COMPONENT VALUE OF U-COMPONENT VALUE OF V-COMPONENT VALUE OF W-COMPONENT VALUE OF W-COMPONENT VALUE OF
FIELD NO.	- CN 4 10 9 V
FIELD LABEL	16 F(16,u) F(16,v) F(16,w) Fdt(16,u) Fdt(16,v)
CARD NO.	0000000 111111 1111111

TIME INPUT LINE. THE ABOVE INPUT IS REPEATED FOR EACH TIME POINT IN THE NOTE:

<u>U</u>
Ć
F
10
U
STRESS CONDITIONS
Ü
110
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INITIAL
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1
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STRES
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<u>-</u>
ı
_
NFC.
불

FORMAT:

;	ZAS
	Z X S
	SXY
	725
	3 × ×
	16:5xx
ISTRES NS	91

EXAMPLE:

	10.0
312.	
<u> </u>	10.0
1	10.0
150.	10.0
2000.	10.0
04 1000.0	02110.0
	04:1000.0 ;2000. ;150. ;300.0 1.0E+4 ;312.0

FIELD DESCRIPTION:

CONTENTS		THE ALPHANUMERICS "ISTRES"
FIELD NO.		<del>न</del> ਜ
FIELD LABEL		ISTRES NS
CARD NO.		

THAT FOLLOW (INTEGER, 1 <ns<28) (INTEGER; 1<ig<28)< th=""></ig<28)<></ns<28) 
THE ALPHANUMERICS "ISTRES"  THE NUMBER OF INITIAL STRESS CONDITIONS THAT FOLLOW (INTEGER, 1 <ns<28) (integer;="" (real)="" (real)<="" 1<16<28)="" component="" for="" grid="" local="" number="" point="" stress="" sxx="" sxy="" sxz="" td="" this=""></ns<28)>
1 ISTRES 1 NS 2-NS+1 IG 2-NS+1 SXX 2-NS+1 SYY 2-NS+1 SXZ 2-NS+1 SXZ 2-NS+1 SXZ

- IDISP - INITIAL VALUES OF DISFLACEMENT

FORMAT:

INPUT

	3.6	
!	. Ddt (IG,w)	
	Ddt (IG, v)	
	Ddt (IG,u)	
	D(IG,w)	
	D(16,v)	
	D(IG,u)	
IDISP ND	16	1

EXAMPLE:

.	0.0	0.0
	0.0	10.0
	-0.00005	-0.00005
	0.0	0.0
021	0310.0	19:0.0
IDISP 02		

FIELD DESCRIPTION:

CONTENTS

FIELD NO.

FIELD LABEL

CARD NO.

THE ALPHANUMERICS "IDISP"	THE NUMBER OF INITIAL DISPLACEMENT VALUES BEING INPUT	LOCAL GRID FOINT NUMBER FOR THIS DISPLACEMENT (INTEGER BETWEEN 1 AND 28)	u-COMPONENT VALUE OF DISPLACEMENT (REAL)	v-COMPONENT VALUE OF DISPLACEMENT (REAL)	W-COMPONENT VALUE OF DISPLACEMENT (REAL)	u-COMPONENT VALUE OF DISPLACEMENT RATE (REAL)	v-COMPONENT VALUE OF DISPLACEMENT RATE (REAL)	W-COMPONENT VALUE OF DISPLACEMENT RATE (REAL)
-	-		ы	M	4	ın	9	7
1 IDISP	1 ND	2-ND+1 IG	2-ND+1 D(IG, w)	2-ND+1 D(IG, v)	2-ND+1 D(IG, W)	2-ND+1 Ddt(IG,u)	2-ND+1 Ddt(IG, v)	Z-ND+1 Ddt(IG,w)

					-						
	İ	•		PSXZ PSYZ;				1			0 = NO PRINT) 0 = NO PRINT) NO PRINT) NO PRINT) NO PRINT) PRINT) PRINT) PRINT) PRINT)
OPTIONS (DEFAILT				PSZZ PSXY			1 0 1				PRINT: PRINT: SINT: O   NO NO O   NO
FRINT				PSXX PSYY			1 1	**			 TACTION ( TRECTION ( TRECTION ( TON (1 = 10N (1 = PRI (1
NT AND STRESS				VELV VELW			0 1			STN	 FRICS IN C IN W IN W U DIRECT V DIRECT NENT ST
-DISPLACEMENT AND				DSFW VELU:	-		1 1:			CONTENTS	 THE ALPHANUM DISPLACEMENT DISPLACEMENT DISPLACEMENT VELOCITY IN VELOCITY IN VELOCITY IN STRESS COMPOSTRESS COMPOST
PDOF -				J DSPV	•			<u>.</u>  -	<u>.e.</u> (	FIELD NO.	1 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2
ſ	<b>)Τ:</b>			DSFU	ר ב:		1	-	 DESCRIPTION:	FIELD	PDOF DSPU DSPW VELU VELV VELW PSXX PSXX PSXX PSXX
INFUT	FORMAT:	-	FDOF		EXAMPLE:	FDOF			FIELD 1	CARD NO.	- 44444444444

		 . <del></del>
j.		
= PRINT AL	-	
(DEFAULT		
CONTROL		 . <u>.</u>
r Print		     
.OCAL GRID NUMBER PRINT CONTROL (DEFAULT = PRINT ALL)		 
LOCAL		 
PGRD -		 •
NPUT -	FORMAT:	 FGRD '

FGRD .				•		• •		•		•	- 1			
	P1 P2		i b	 F4	P.S.		F7		64	P10:	F11	P12:	P13	F14
	P15	! P16	P17	P18(	F19	P20:	P21	P 22	P23	P24	P25	P26	P27	P28
	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1						1111111		1 1 1		 	! !		

EXAMPLE:

								(		
	1 1: 1	 1		1 1 1			 1	-	1 1 0 0 1 1	1 1
1 11 1	1	 	0	1	-	0	 =	·	1 1	 +

FIELD DESCRIPTION:

CARD FIELD FIELD CONTENTS		FIELD NO.	FIELD	CARD NO.
---------------------------	--	--------------	-------	-------------

THE ALPHANUMERICS "PGRD" BE FRINT CONTROL FOR LOCAL GRID FOINT NUMBERS 1 - 14 (1 = PRINT; O = NO PRINT) BE PRINT CONTROL FOR LOCAL GRID POINT NUMBERS 15 - 28 (1 = PRINT; O = NO PRINT)
" " 55
14 28
1 1
15
NUMBERS NUMBERS
LNIO LNIO
9 0
GRI
"PGRD" LOCAL LOCAL
20 G 80 A 80 A
流 도 교
A P N C P N
A TIN
+ # #
w w
411
38.2
F (14
FGRD 1 F(1)-F(14) 2A - F(15)-P(28)2A -
0 0 0
<b>→ 0</b> 10

NOTE: INTEGER INPUT REQUIRED. RIGHT JUSTIFY.

INPUT	1	CND	ICND - ROUNDED	GRID	POINT	IDENTIFIC	CATION	(DEFAUL	(DED GRID POINT IDENTIFICATION (DEFAULT = NONE BOUNDED (1)(1)	JUNDED	( T( )(
FORMAT:					•					<b>!</b>	
1-											
	N(1) N(2) N(3)	N (2)	(M) N								1
	N(15)			- (EN) N							
							}				
EXAMPLE:											
		<u>.</u>     		; 			,				
ICND 17				    -  -	       		<del> </del>				į
	-	111	!	21 1	16 13	20	27!	28 17	5	- F	14
			18		1 1 1					-	
		-					•		_		

FIELD DESCRIPTION:

CONTENTS

FIELD NO.

FIELD LABEL

CARD NO.

ERICS "ICND" F BOUNDED GRID POINTS (LOCAL) IDENTIFIED ON LINES 2 OR 2 AND 3	(INTEGERS BETWEEN 1 AND 28)
N N N N N N N N N N N N N N N N N N N	LOCAL GRID POINTS BOUNDED OUT OF PROBLEM (INTEGERS BETWEEN'I AND LOCAL GRID POINTS BOUNDED OUT OF PROBLEM (INTEGERS BETWEEN'I AND
ICND 1 NA 1	N(1)-N(14) 2A - BB N(15)-N(NB)2A - BB
	NN

PROBLEM EXECUTION CONTROL PARAMETERS (REGUIRED)		DELTM : :		CONT :1002 :.05 :.0E-04 !
- PROBLEM EXEC				05
- CONT -		CONT MiPC CRITR		1002
INFUT	FORMAT:	CONT	EXAMPLE:	CONT

FIELD DESCRIPTION:

CARD NO.	FIELD LABEL	FIELD NO.	FIELD CONTENTS NO.  1 THE ALFHANUMERICS "CONT"
	Ωi	И	THE MAXIMUM NUMBER OF ITERATIONS TO EXECUTE; DEFAULT TO 3 IF RANGE EXCEEDED.
-	ņ	и	(INTEGER BEIWEEN I AND 20) PRINTED DUTPUT FORMAT (INTEGER; 1 OR 2) DEFAULT = 1 1 = PRINT GRID FOINT VALUES AT ALL TIMES
-	CRITR	ю	2 = PRINT ALL GRID POINT VALUES FOR THIS TIME THE CONVERGENCE CRITERIA FOR ALL DISPLACEMENT VALUES; RELATIVE CONVERGENCE,
	DELTM	4	DEFAULT TO 0.01 (REAL > 0.) PRINTED OUTPUT TIME INCREMENT

INFUT - END (REQUIRED LAST INFUT)

FORMAT AND EXAMPLE:

#### 6.0 OUTPUT OVERVIEW

Typical output is illustrated in Section 7, Figure 5. An echo print of the user input is always printed first. The echo print is followed by a group of program controlled output which contains:

- 1. Computed local coordinates of analysis element
- 2. Labeled material properties, as input
- 3. Tables giving element load values in the local coordinate system for each input time slice
- 4. Table of input initial stress and displacement values. The third data output grouping is output each iteration and contains the displacement/stress data computed that iteration. This output is only at the input time slices. This data is not present for the LEQP case since only one iteration is performed.

The final output consits of displacement/stress data at the requested print time increments upon convergence of the problem. The printed format is operator controlled for this grouping.

# 7.0 OUTPUT EXAMPLE

A quadrilateral plate of constant thickness is sued to illustrate the input-output format of SFENES. Table 1 provides the required input to solve a linear quad plate problem and Figure 4 shows the geometric layout of the problem plate. Used input data is illustrated in Figure 3 and each input record is described in Section 5 of this report.

Problem output is illustrated in Figure 5. A description of this output is contained in Section 6 of this report. The output contrived here is only partial (unbounded D.O.F. only) in order to conserve space.

QUADRILATERAL PLATE MATERIAL PROPERTIES, COORDINATE INPUT,

LOAD EQUIPMENT AND INITIAL VALUES

# (1) Material Properties

 $G = 1.1538 \times 10^7$ 

 $\rho = .2588 \times 10^{-3}$ 

 $\mu = .30$ 

# (2) Spatal Coordinates

Grid No.	X	Y	Z	Thickness
13	0	0	0	1.0
14	10.0	10.0	0	1.0
15	10.0	10.0	0	1.0
16	0.0	10.0	0	1.0

# (3) Time Coordinates

0.0

0.0005

0.0010

# (4) Element Load

Pressure Load on Top Surface of

-10,000t or -10,000t in W dir

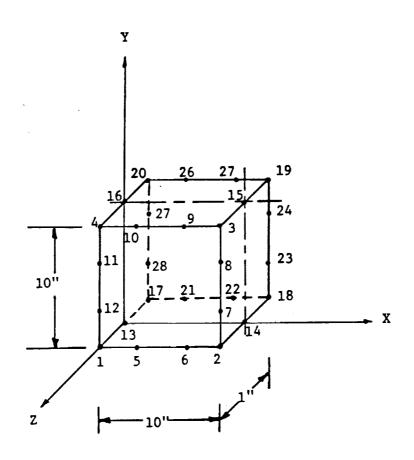
#### (5) Initial Values

Stress = 0.0 at all grids

Displacement = 0.0 at all grids

FIGURE 4

QUAD PLATE DIMENSIONS



LOAD AT GRID POINTS 1-12

DIRECTION

+Z

VALUE

-10,000t

FIGURE 5.

<del></del>		<u> </u>	2 2 2 2	2 2 2 2	<u> </u>	<u> </u>	1 0 F F	<u> </u>	7. TR # 6	<u>इंच</u> इंच	<u> </u>	9	<u>នាទីន</u>		<b>. 11</b>	2 4 5 2	111	7777
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	1E-03					1		16										
	0.2588001E-03		-0.1030300E+05	0.0	0.00	1		14										
E-03		E+01 E+01	•			1		13										
-999999€-03	300000E+00	1000000E+01	0 0	0.0	0.0	1		12							:		i	
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0.1000000E+02 0.1000000E+02 0.1000000E+02	0.3333335+01 0.1000000E+02 0.3333335+01	0.5030000E+00 0.0 -0.5900000E+00	
0.6666666E+01 0.1000000E+02 0.666666E+01	0.0	0.5000000E+00 0.0 -0.500000E+00	
0.3333333E+01 0.0 0.33333333E+01	0.0	0.5000000E+00 0.0 -0.5000000E+00	0
0.00	0.100000UE+02 0.3333334E+01 0.1000000E+02 0.3333334E+01	0.5000000E +00 0.5000000E +00 -0.5000000E +00 -0.5000000E +00 0.1000000E +00	0
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0.10000000F+02 0.3333334E+01 0.1000000E+02	0.0 0.1000000E+02 0.0 0.1000000E+02	0.5000000E+00 0.500000E+00 -0.500000E+00 -0.500000E+00	2
x-COORD 0.0 0.6666667E+01 0.0 0.4666667E+01	Y-COORD 0.0 0.1000000=02 0.0 0.10000000E+02	0000E+00 0000E+00 0000E+00 000CE+00 TPICKNESS 0CCOE+01	AMU 0.300000000000000000000000000000000000
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<del>নিন্ন্নিক্তাই লাহত এছা হোটাৰ</del> এই হ<sub>ো</sub> বিষয়ে এই বিষয়ে এই স্থানিক স্থানিক স্থানিক ইন্নানিক বিষয়ে হোটাৰ হো ) 0.0 0.1000000E-03 0.2000000E-03 0.399998E-03 0.599999E-03 0.599999E-03 0.8999999E-03 

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<u>েন্ন্--ন্ন্ন্তট ল্লেট্টছাল্টলয়ন্ধ্রিস্থ টো গায় স্ট্রাধ্র সংখ্য চল্লেল এখন এখন এখন স্বর্টসাল করি স্থার ইয়ার ইটিয়ার স্ট্রার স্ট্রার ইটিয়ার স্ট্রার ইটিয়ার স্ট্রার স্ট্</u> -0.4080325E+00 -0.1898656E+CO -0-3258761F+00 -0.4363334E+00 -0.2530695E+00 -0.1593109E+00 -0.1636479E+00 -0.2617448E+00 0 0.5480688E-01 0.3363953E-01 0.9027678E-01 0.9709895E-01 0.4024063E-01 0.4172725E-01 0-9228629E-C1 0.6800932E-01 0-7181942E-0 -- VEL DC 17Y---0.3939573E-02 0.7232644E-02 0.1498196E-01 0.1636703E-01 0.8395758E-02 0.6214779E-02 0.1439830E-01 0-1101420E-01 ∍ -0.1159891E-03 -0.1587938E-03 -0.7332212E-04 -0.1914533E-03 -0.2115394[-03 -0.2395718E-93 -0.2603086E-03 -0.9942053E-05 -0-3617797E-04 -0-2254470E-03 3 0.5299707E-04 0.5827364E-04 -- DI SPIACEMENT-0-1615938E-04 0.2562509E-04 0.4669650E-04 0.21 8332 DE -05 0-7957626E-05 0.4241010E-04 Da4 9681 74E -04 0.0 0.6936333E-05 0.7519157E-05 0.7859864E-05 0.2478196E-05 0.4028253E-05 0.9442016E-05 0.32253296-06 0-11 95777E-05 0.8365192E-05 > d 0.0 0.5999999E-03 0.7000000E-03 0.3999998E-03 0.300000E-03 0.8999999E-03 0.1000000E-03 E 0.0 

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<u>त्रोम् १९४५ वर्षे स्वार्थे स्वत्राहर्ते । स्वत्राहर्ते स्वत्राहरू । स्वत्राहरू स्वत्राहरू स्वत्राहरू स्वत्र</u>ाहरू 0.5899037E-01 -0.6609318E+00 0.2911801E-01 -0.3797050E+00 0.6449628E-01 -0.5573416E+00 -0.5085227E+00 -0.8816684E+00 0.1062498E+00 -0.1119441E+01 0.1190613E+00 -0.1221836E+01 0.3095343E-01 -0.3451751E+00 0.1297473E+00 -0.1016213E+01 0 0.4671912E-01 --VELOCITY 0.0 0.5130607E-01 0.6939816E-01 0.5887758E-01 0.616B39BE-0 0.2796507E-01 0.11977025-01 0.1341994E-01 0.3259344E-01 0.2186724E-01 0.3896923E-0 5 ŧ 0.0 -0.7325206E-03 -0.9719204E-04 -0.1983755E-03 -0.3165670E-03 -0.4332308E-03 -0.5286639E-03 -0.5786400E-03 -0.6123282E-03 -0.6558986E-03 -0.2655422E-04 0.0 0.0 0.7146665E-04 0.1848052E-04 0.2984023E-04 0-4191609E-04 0.5062987E-04 0.5477107E-04 0.5751041E-04 0.6201869E-04 0.2430141E-05 0.8967078E-05 -- DI SPLACEMENI 0-421460BE-05 0.8768077E-05 0.1431655E-04 0-2038477E-C4 0.2472027E-04 0.2768337E-04 0-2983715E-04 0-2655545E-04 0.34789A2E-C4 0.1133070E-05 > 0:0 ŧ 0.5999996-03 0.7000006-03 0.8000006-03 0.89999996-03 0.99999996-03 0.399998E-03 0.500001E-03 0.1000000E-03 TIME 0.0 0.0 22228282 F F 9 F 2 7 7 7 7 7 8 8 8 2 2 2 2 2

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GRID NO. 11

<u>েন্ন্ন্ন্ৰ্বিটাই বাহৰ বৰ্ষ বিশ্ব নাম ১৯ খন চল্পান লগান কৰিব লগান কৰিব লগান কৰিব বাহৰ চৰ্মান কৰিব লগে কৰিব লগান</u> --OT SPLACEMENT l 5 0.0 0.10000000E-03 0.3000000E-03 0.399998E-03 0.599999E-03 0.5999999-03 0.8999999E-03 0.8999999E-03 TIME 

GR 10 NO. 12

<u>েন্দ্ৰীৰ বিষ্টাৰ চাইছিল লাল লাল প্ৰাৰ্থ কৰাৰ প্ৰথম গাল্ভ লাল কৰি প্ৰাৰ্থ কৰি প্ৰাৰ্থ লাল লাল লাল কৰিব লাল ই ব</u> 1 FLOCITY-00000000000 1 00000000000 0200000000 d 0.0 0.1000000E-03 0.200000E-03 0.399998E-03 0.599999E-03 0.599999E-03 0.899999E-03 0.899999E-03 TIME n n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n | p n |

GRID NO. 13

<u>তৰ আলমৰ প্ৰাণ্ড লাগ সভ্যাৰ প্ৰাণ্ড মান্ত মান্ত ৰ লাগৰে লাগৰ মুচ পামৰ মুখা আৰু ৰ ভাগ অনু ৰাজ চুট আই চুট মান্ত মুখ</u> ı 3 d -DISPLACEMENT-.... 0.0 0.1000000E-03 0.2000000E-03 0.399958E-03 0.599999E-03 0.7000000E-03 0.8999999E-03 0.999999E-03 T IME

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<u>হত্ত্বাল্য লোক্ষ্ণ স্থান কৰ্ম কৰি কৰ্ম বিশ্ব ক্ষাৰ্থ কৰিছে এই প্ৰথম কৰিছে এই আছে আই বৃহত্ত্বাল্য ই শ্বন্ধ কৰিছ</u> • • • • • -0.1889451E+00 -0.3246374E+C0 -0.1781030E-01 -0.4122002E+00 -0-2499249E+00 -0-1281928E+00 -0.4070777E+00 -0.4362649E+00 -0.1552559E+00 -0.1687350E+Q0 -0.2768841E+00 0 -0.1069558E-01 -0.1352314E-01 -0.6556831E-02 -0.1619898E-02 -0.5453363E-02 -0.1613952E-01 -0.2470826E-01 -0.9961065E-02 1 -- YEL OC 1 TY--0.5421118E-01 -0.9699136E-01 -0.8998907E-01 -0.2855689E-01 -0.4118775E-01 -0.7079101E-01 -0.4149340E-01 -0.7148969E-01 -0.3289798E-01 ∍ -0.7304341E-04 -0-1585215E-03 -0.2603696E-03 -0.9891000E-JS -0.3601394E-04 -0.1156542E-03 -0.1913643E-33 -0.2107600E-03 -0.2386520E-03 -0.2243689E-J3 -0-2241240E-05 0.5999999E-03 -0.4229420E-04 -0.6603347E-05 -0.9487551E-05 -0.3732250E-05 -0-7137876E-05 -C.1059150E-05 0.5000001E-03 -0.3510027E-C4 -0.5437630E-05 -0.6959547E-05 --DISPLACEMENT--0.1608518E-04 -0.25 52998E-04 -0.46 53823E-C4 -3.4943954E-04 -0.5278536E-04 -0.5824286E-04 -0-21 704 76E-05 -0.7915451 0.399998E-03 0.1000doof-03 0.2000000E-03 0.3000000E-03 0.7000000E-03 0.899999E-03 0.9999999E-03 TIME 8 2 2 2 2 2 2 2 2 2 2 2 2 1 1 1 1

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۱ 0.7730046E+02 -0.3188586E+02 -0.9735374E+01 0.1310131E+03 -0.6523181E+02 -0.2355817E+02 ; SXZ 0.6657388E+02 0.1193960E+03 275 0.1972791E+03 0.3441958E+03 SYY 0.19861 08E + 02 0.3474673E + 02 SXX 0.5000001E-03 T IME 

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